

Notes by-

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→ No bending of member

→ Truss : Compression member \Rightarrow strut

Building : Colm or stanchions.

Crane : Boom.

Main comp. member in roof truss : Rafter.

$\sigma_{ac} = 80 \text{ MPa} \rightarrow$ General member

$\sigma_{ac} = 110 \text{ MPa} \rightarrow$ Heavy (Built up)

$\sigma_{ac} = 150 \text{ MPa}$

20 mm pd. Rivet $\left\{ \begin{array}{l} \sigma_{lf} = \text{Bending strength} = 300 \text{ MPa} \\ \sigma_{vt} = \text{Shearing strength} = 100 \text{ MPa} \end{array} \right.$

- * comp. member rarely fails by crushing.
- * Per. comp. strength is minimum for slender colm; which can easily buckle under the less compressive load.

* Strength of compression member :- [Axially loaded]

The max. axial compressive load 'P' which can be permitted on a comp. member is given by,

$$P = \sigma_{ac} \cdot A$$

Where, P = axial comp. load (N)

σ_{ac} = per. stress in axial comp. (MPa)

A = eff. c/s area of member. (mm^2)

= Gross c/s area - Deduction for holes not filled completely by rivets or bolts.

σ_{ac} can be calculated by [Merchant Rankine formula - Cl. 5.1.1]

or Table 5.1, Cl. 5.1.1, gives σ_{ac} depending on f_y & $[A = l_e/r_{min}]$

* Effective length - Effective length is the length betw' point of contraflexure ; & when accurate frame analysis is not done, effective length can be directly obtained from [Table 5.2 / Cl. 5.2.2 / Pg. 41] depending on end conditions.

* Angle struts :- [Cl. 5.5 / Pg. 46]

for angle struts, design specifications are slightly different.

Table on next page gives effective length, allowable stress & max. slenderness ratio for angle comp. member depending upon end conditions.

Types of angle strut :-

① Discontinuous

Double angle, any No. of rivet

or bolts or weld on both sides of GP

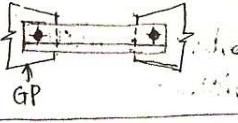
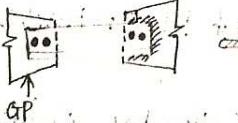
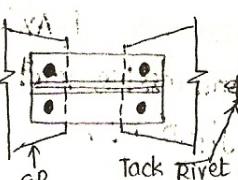
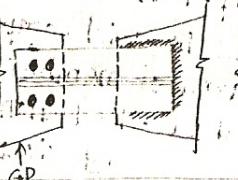
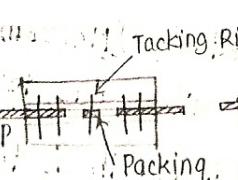
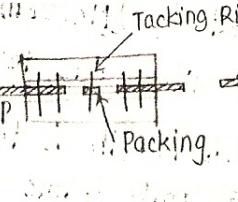
single angle & single angle, one Rivet, two Rivet / Weld at each end at each end

Double angle, any No. of rivet, weld, placed on same side of GP

One No. of rivet or bolt, rivet or bolt or weld at each end at each end

single angle double angle, one Rivet, two Rivet / Weld at each end at each end

One No. of rivet or bolt, rivet or bolt or weld at each end at each end

Sr. No.	Type	End condition	eff. length	allowable stress
①	Single angle Discontinuous	a) one rivet or bolt at each end. b) Two or more rivets or bolts or welding at each end.	 	$l_e = L$ $l_e = 0.85L$
②	Double angle discontinuous [Tacking Provided]	a) connected on side side of GP i) One rivet or bolt at each end. ii) Two or more rivets, bolts or weld at each end.	 	$l_e = L$ $l_e = 0.85L$
		b) connected on both sides of GP by two or more rivets or bolts or welding.	 	$l_e = 0.7 \text{ to } 0.85L$ (Depending on rigidity of joint)
③	Single or double angle Continuous	one or more rivet or bolt or welding.		$l = 0.7L \text{ to } 0.85L$ (Depending on end rigidity)

* Truss: In case of bolted, riveted or welded trusses, in braced frames, the effective length 'l' of comp. member is taken in betⁿ 0.7L to 1L (where L = dist. betⁿ intersection) depending upon degree of restraint [end conditions] provided.

In case of truss, buckling in the plane of truss \perp or \parallel to members of the plane of truss, eff. length is taken as 1.0 times dist. betⁿ points of restraint.

(GATE) Max. slenderness ratio: Slenderness Ratio = $\lambda = \frac{L}{r_{min}}$ should not exceed: \rightarrow

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Sr. No.	Member or superimposed load	$\lambda_{max.}$
①	Member carrying comp. load (DL+LL)	180
②	Tension member; Reversal of stresses due to load other than WL & EQ loads	180
③	Comp. member due to WL & EQ load (provided that deformat' of such member does not adversely affect the stress in any part of the str.)	250
④	Comp. flange of a beam	300
⑤	Tie member in roof truss or Bracing system [subjected to possible reverse of stress due to WL & EQ load]	350
⑥	Tension member (except - pretensioned)	400

* Compression member :- Two components Back To Back :- [Cl. 5.9 & 10.8-10]
IS: 800-1984 Cl. 8.10 and 5.9] Recommendations:-

① Tacking Rivets:- A comp. member composed of two angles, channels, or tees back to back in contact or separated by a small distance should be connected by riveting, bolting or welding, so that the slenderness ratio (L/e_{min}) of each member bet' connections is not greater than 40 & nor - greater than 0.6 times most unfavourable slenderness ratio of the whole struc.

i.e. $\lambda \neq 40$
 $\neq 0.6 \times \text{unfavourable.}$ } For comp. member - back to back with (tacking $\neq 600 \text{ mm}$)

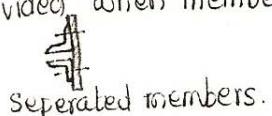
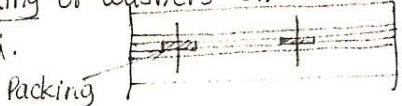
For other types of "built up compression member" say, where cover plates are used; pitch of tacking rivets should not exceed $32t$ or 300mm , whichever is less. (where $t = \text{Thk. of thinner outside plate}$). Where plates are exposed to weather [Due to weathering effect - outside pitch is decreased], pitch should not exceed $16t$ or 200 mm whichever is less.

i.e. $\boxed{\text{Pitch} \neq 32t \text{ or } 300\text{ mm}} \rightarrow \text{Built up comp. member, No weathering}$
 $\neq 16t \text{ or } 200\text{ mm} \rightarrow \text{Subjected to weathering - outside}$

* Min. dia. of Rivets:-

Thk. of member	min. dia. of rivet	Bolt Rivet weld	Should carry SF & BM if any
$< 10 \text{ mm}$	16 mm		
$10-16 \text{ mm}$	20 mm		
$> 16 \text{ mm}$	22 mm		

* Solid packing or washers should be provided when members are separated.



- ② The ends of struts should be connected together with not less than 2 rivets or bolts or their equivalent in welding & there should not be less than 2 additional connections spaced equidistant in the length of strut.

* Min. 2 Rivets or bolts used at end connection one on line each gauge mark } Where
 Legs of connected angles
 or flange of connected Tee $> 125\text{mm}$ wide
 or to web of channel connected $> 150\text{mm}$ wide.

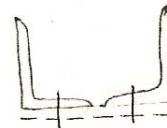
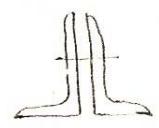
- * Typical cross-section of compression member:-

$$\text{Permissible comp. stress} \propto \frac{1}{\text{slenderness Ratio}} \propto \frac{r_i}{l_e}$$

∴ section should be proportioned to have max. radius of gyration.

GATE * circular pipe has best c/s in this regard & have highest radius of gyration in all directions. However due to difficulties in joining these sections are not preferred.

Various c/s used for compression member are:-

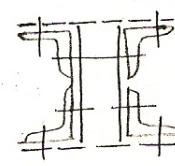
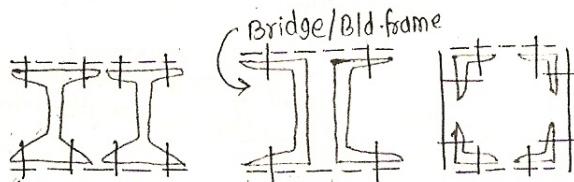
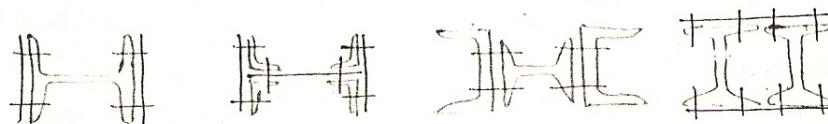


Single angle section
 ⇒ simplest type
 ⇒ Used in trusses,
 ⇒ Bracing in PG
 ⇒ Built up colm.

Not economical as
 r is min. in minor principal
 axis & connect' causes eccentricity.

I section & sc sections
 are most suitable.
 as diff. betw 'r' @
 two axes is smallest

--- → Lacing
 ——— cover plate



Dotted line ⇒ --- → Lacing
 Thick line ⇒ ——— → cover plate.

Prob] A column consisting of sc. 140 @ 33.3 kgf/m has an unsupported length of 4m. It is effectively held in position & restrained against rotation at one end & at the other end restrained against rotation but not held in position.

Determine the axial load this colm can carry if the steel conforms to IS 226-1975.

Solⁿ: For SC: 140 @ 33 \rightarrow fy = 250 MPa
 $a = 4.24 \times 10^3 \text{ mm}^2$
 $r_x = 58.9 \text{ mm}$
 $r_y = 32.1 \text{ mm}$

f_y = Yield stress = 250 MPa - for steel conforming to IS 226-1975
 \therefore Slenderness ratio = $\frac{\lambda e}{r_{\min}}$.

From IS: 800-1984; Pg - 41/c1.5.2.2, Table 5.2;
 Effective length = $\lambda e = 1.2L \rightarrow$ case (d)
 $= 1.2 \times 4$
 $= 4.8 \text{ m}$.

$$\therefore A = \frac{4.8 \times 10^3}{32.1} = 149.53.$$

From Table 5.1/Pg. 39, & for $\lambda = 149.53$ & f_y = 250 MPa.

λ	δ_{ac}
140	51
150	45
149.53	?

$$\therefore \text{By interpolation; } \delta_{ac} = 51 - \frac{(51-45)}{(150-140)} (149.53-140)$$

$$\underline{\delta_{ac} = 45.282 \text{ MPa.}}$$

$$\therefore \delta_{ac} = \frac{P}{A} \Rightarrow P = A \cdot \delta_{ac} = 4.24 \times 10^3 \times 45.282$$

$$\boxed{P = 192 \text{ kN}}$$

Pro 2] A single angle strut ISA 50x50x6 mm of a roof truss is 1.06 m long.
 It is connected by one rivet at each end. Determine the safe load
 this strut can carry.

Solⁿ: For ISA 50x50x6 mm;

$$A = 5.68 \times 10^2 \text{ mm}^2$$

$$I_{xx} = 12.9 \text{ cm}^4 \quad I_{yy} = 5.3$$

$$C_{xy} = C_{yy} = 1.95 \text{ cm}, \quad \therefore r = \sqrt{\frac{I}{A}} = \sqrt{12.9 \times 10^4 / 5.68 \times 10^2} = 0.966 \text{ cm} = 9.66 \text{ mm}$$

$$\lambda e = 1.06 \text{ m.}$$

$$\lambda = \frac{\lambda e}{r_{\min}} = \frac{1.06 \times 10^3}{9.66} = 109.73$$

From Table 5.1/Pg. 39

λ	δ_{ac}
100	80
110	72

$$\therefore \delta_{ac} = 80 - \frac{(80-72)}{(110-100)} (109.73-100)$$

$$= 72.216 \Rightarrow \text{see table,} \\ \underline{\text{permissible stress} = 0.8 \times \delta_{ac}}$$

$$\therefore \delta_{ac} = \frac{P}{A} \Rightarrow P = \delta_{ac} \times A \\ = 72.216 \times 5.68 \times 10^2 \times 0.8 \\ \underline{\underline{10^3}} \\ = 44.05 \text{ kN.} \underline{32.81 \text{ kN.}}$$

for discontinuous single
 angle strut connected by
 single rivet at end.

Prob: A double angle discontinuous strut consist of 2 ISA-6040, 6 mm thk. with longer legs back to back connected to opposite sides of a GP 10mm thk. with two rivets. The length of strut betn c/c of intersection is 4m. Assuming tacking rivets have been provided along the length, determine the safe load carrying capacity of the section, if the yield stress for steel = 250 MPa.

Sol:- For 2 ISA 60x60x6 mm, from steel table

$$a = 6.84 \text{ cm}^2 \Rightarrow 2a = 13.68 \text{ cm}^2$$

$$I_{yy} = 9.1 \text{ cm}^4 \Rightarrow 2I_{yy} = 18.2 \text{ cm}^4$$

$$r_x = 1.88 \text{ cm}$$

$$c_y = 1 \text{ cm}$$

$$I_y = 7 \text{ cm}^4 \quad 39.425 \text{ cm}^4$$

$$\therefore r_{min} = \sqrt{I/a} = 0.87$$

$$L = 4 \text{ m}$$

$$\therefore r_y = \sqrt{I_y/a} = \sqrt{7/5.65} =$$

for two angle discontinuous strut with 2 rivets at each end,

$$l_e = 0.85L \quad \text{& Per. stress in comp} = 6ac.$$

From steel table, for ISA 60x6, 6 mm thk,
 $a = 5.65 \text{ cm}^2, r_x = 1.88 \text{ cm}, c_y = 1.0 \text{ cm}, I_y = 7 \text{ cm}^4$

$$\text{for strut, } r_x = 1.88 \text{ cm}$$

$$I_y = 2 \times 7 + 2 \times 5.65 (1+0.5)^2 = 39.425 \text{ cm}^4$$

$$\therefore r_y = \sqrt{\frac{I_y}{A}} = \sqrt{\frac{39.425}{2 \times 5.65}} = 1.87 \text{ cm}$$

$$\therefore \text{Min. radius of gyration} = r_{min} = r_y = 1.87 \text{ cm}.$$

$$l_e = 0.85L \rightarrow \text{from table}$$

$$= 0.85 \times 4$$

$$= 3.4 \text{ m}$$

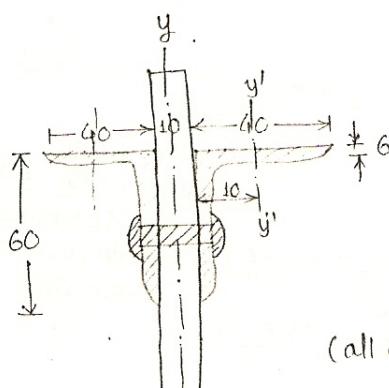
$$\therefore \text{Max. slenderness ratio} = \lambda = \frac{l_e}{r_{min}} = \frac{3.4}{1.87} = 181.8$$

For strut used to carry load due to WL & EQ load, $\lambda_r > 180$

\therefore This strut should not be used for the same.

$$\therefore \text{From table 5.1, } \delta_{ac} = 33 - (33-30) \times \frac{1.8}{10} = 32.46 \text{ MPa}$$

$$\therefore \text{Safe load} = 2 \times 32.46 \times 5.65 \times 100 = \underline{36.67 \text{ kN}}$$

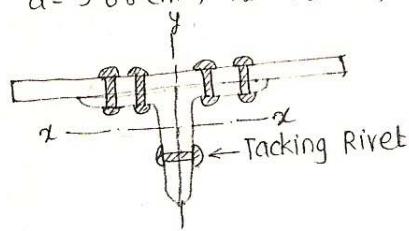


(all dim. in mm)

$$\begin{aligned} I_y &= \frac{1}{3} I_y' + A \cdot b^2 \\ &= 2 \times 7 + 2 \times 5.65 \times (1+0.5)^2 \\ &= 39.425 \text{ cm}^4. \end{aligned}$$

Prob: 4) A discontinuous strut 1.75 m long (effective) consist of two equal angles ISA 50x50x6 mm. It is connected to the same side of GP by two rivets on each angle at both ends. calculate the load which this strut can carry if yield stress of steel is 250 MPa.

Soln: I] Section Properties: from steel table: for ISA 50x50x6 mm,
 $a = 5.68 \text{ cm}^2, r_x = 1.51 \text{ cm}, I_y = 12.9 \text{ cm}^4, c_y = 1.45 \text{ cm}$



$$\therefore I_y = 2 \times 12.9 + 2 \times 5.68 \times (1.45)^2 = 49.68 \text{ cm}^4.$$

$$\therefore r_y = \sqrt{\frac{I_y}{a}} = \sqrt{\frac{49.68}{2 \times 5.68}} = 2.09 \text{ cm}$$

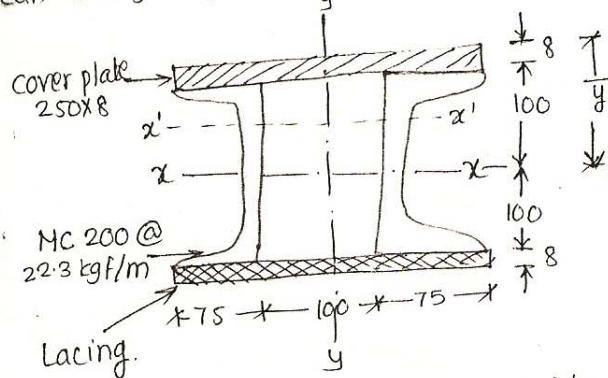
$$\therefore r_{\min} = r_y = 1.51 \text{ cm.}$$

$$\text{II] Slenderness Ratio } = \lambda = \frac{l_e}{r_{\min}} = \frac{1.75 \times 10^2}{1.51} = 115.89 \quad \text{Table 5.1.}$$

$$\text{III] Permissible compressive stress } = \sigma_{ac} = 72 \div (72 - 64) \times \frac{5.89}{10} = 67.29 \text{ MPa.}$$

$$\text{IV] Load carrying capacity } = P = \sigma_{ac} \times a = \frac{2 \times 67.29 \times 568}{1000} = 76.44 \text{ kN}$$

Prob: 5] The top chord of a bridge buss has an effective length of 4m & c/s as shown in fig. Determine the comp. load, this member can carry if yield stress of steel is 250 MPa.



① Section properties of MC 200 @ 22.3 kgf/m from steel table,
 $a = 28.5 \text{ cm}^2, c_y = 2.18 \text{ cm}, I_x = 1830 \text{ cm}^4, I_y = 141 \text{ cm}^4$

② For Strut:

$$\text{c/s area} = 2 \times 28.5 + 25 \times 0.8 = 77 \text{ cm}^2.$$

ΣM @ top fibre to calculate \bar{y} ,

$$\bar{y} = \frac{(2 \times 28.5)(10+0.8) + 25 \times 0.8 \times 0.4}{2 \times 28.5 + 25 \times 0.8} = 8.098 \text{ cm.}$$

$$\therefore M.I \text{ of section} = I_x = 2 \times 1830 + 2 \times 28.5(10.8 - 8.098)^2 + \frac{0.5 \times 0.8^3}{12} + 25 \times 0.8 (8.098 - 0.4)^2$$

$$= 5262.39 \text{ cm}^4$$

$$I_y = 2 \times 141 + 2 \times 28.5 (5 + 2.18)^2 + \frac{0.8 \times 25^3}{12}$$

$$= 4262.15 \text{ cm}^4.$$

as $I_x > I_y$; $r_x > r_y$

$$\therefore r_{\min} = r_y = \sqrt{\frac{I_y}{A}} = \sqrt{\frac{4262.15}{2 \times 28.5 + 25 \times 0.8}} = 7.44 \text{ cm.}$$

$$\textcircled{2} \therefore \text{Max. slenderness ratio} = \lambda_{\max} = \frac{l}{r_{\min}} = \frac{400}{7.44} = 53.76.$$

$$\textcircled{4} \text{ Permissible comp. stress} = \delta_{ac} = 132 - (132 - 122) \times \frac{3.76}{10} = 128.24 \text{ MPa.}$$

$$\textcircled{5} \therefore \text{Safe load carrying capacity of strut} = \delta_{ac} \times a \\ = \frac{128.24 \times (2 \times 28.5 + 25 \times 0.8) \times 100}{1000} \\ = \underline{\underline{987.448 \text{ kN}}}$$

* Design of Compression member:-

① Assume $\delta_{ac} \leftarrow \begin{cases} 80 \text{ MPa} \\ 110 \text{ MPa} \end{cases} \rightarrow \text{single angle-channel-I section} \right.$

② Find $A_{reqd} = \frac{\text{Axial Comp. load}}{\text{Per. comp. stress}} = \frac{P}{\delta_{ac}}$

③ choose trial section having $A \approx A_{reqd}$.

④ Determine λ_{\max} & hence $\delta_{ac \text{ act}}$.

⑤ Safe load carrying capacity = $\delta_{ac \text{ act}} \times A_{pro} \geq P$

.. O.K. otherwise Revise the section.

⑥ check for slenderness ratio according to IS: 800-1984, Table 3.7.1 / Pg 30

Pro 1] Design a column section to carry axial load of 410 kN. The column is 4.2 m long & is effectively held in position at both ends but restrained against rotation at one end only.

Yield stress of steel = 250 MPa.

Soln:- ① Assume $\delta_{ac} = 80 \text{ MPa}$.

$$② A_{reqd} = \frac{P}{\delta_{ac}} = \frac{410 \times 10^3}{80} = 5125 \text{ mm}^2$$

③ From steel table, try SC-160 @ 41.9 kgf/m

Cross area = $a = 5340 \text{ mm}^2$

$r_{\min} = 36.1 \text{ mm.}$

$$④ \text{Max. slenderness ratio} = \lambda_{\max} = \frac{l_e}{r_{\min}} = \frac{0.8 \times 4.2 \times 10^3}{36.1} = 93.07 < 180$$

$$⑤ \therefore \delta_{ac} = 90 - (90 - 80) \times \frac{3.07}{10} = 86.93 \text{ MPa}$$

[From IS: 800 / Table 5.1]

$$⑥ \therefore \text{Safe load carrying capacity} = \frac{86.93 \times 5340}{1000} = 464 \text{ kN} > 410 \text{ kN}$$

∴ O.K.

Use SC 160 @ 41.9 kgf/m

Pro: 2] Design a suitable angle strut for a roof truss carrying a comp. load of 97 kN. The length of strut bet' c/c intersection is 910 mm. Also design a) Riveted b) Welded connections.

Soln:- ① Assuming for strut that both ends are connected by more than one rivet & yield stress of steel = 250 MPa.
 $\therefore d_e = 0.85L = 0.85 \times 2100 = 1785 \text{ mm}$.

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② Assume $\delta_{ac} = 80 \text{ MPa}$

$$A_{reqd} = \frac{P}{\delta_{ac}} = \frac{97 \times 10^3}{80} = 1212 \text{ cm}^2$$

③ Try for single angle ISA 80X80X8 mm

$$a = 12.21 \text{ cm}^2, r_{min} = 1.55 \text{ mm}$$

④ \therefore Max. slenderness ratio = $\lambda_{max} = \frac{l}{r_{min}} = \frac{1785}{1.55} = 115.16 < 180$
 \therefore O.K.

⑤ \therefore from Table 5.1 / Pg 39, $\delta_{ac} = 67.87 \text{ MPa}$.

⑥ \therefore Load carrying capacity = $\delta_{ac} \times a = 67.87 \times \frac{97 \times 10^3}{1000} 1212 = 82.87 \text{ kN} < 97 \text{ kN}$

\therefore Unsafe.

⑦ Revise the section.

Let $\delta_{ac} = 68 \text{ MPa} \rightarrow$ from step ⑤

$$\therefore A_{reqd} = \frac{97 \times 10^3}{68} = 1426 \text{ mm}^2$$

⑧ Try for single angle ISA 80X80X10 mm,
 $a = 1505 \text{ mm}^2, r_{min} = 1.5 \text{ mm}$

⑨ $\therefore \lambda_{max} = \frac{l}{r_{min}} = \frac{1785}{1.5} = 118.66 < 180$
 \therefore O.K.

⑩ \therefore from Table 5.1 / Pg. 39, $\delta_{ac} = 67.87 \text{ MPa}$.

⑪ \therefore Safe load carrying capacity = $\frac{67.87 \times 1505}{1000} = 102 \text{ kN} > 97 \text{ kN}$
 \therefore O.K.

a) Design of Riveted end connection:

① Strength of 18 mm dia. shop rivets

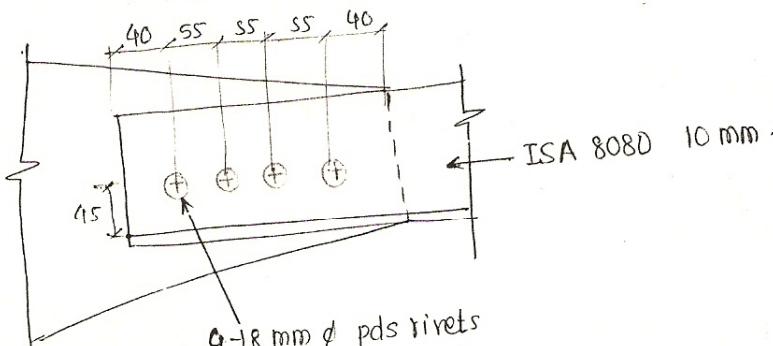
$$\begin{aligned} \text{In single shear} &= 100 \times \frac{\pi}{4} \times 19.5^2 \\ &= 29.86 \text{ kN} \\ \text{In bearing } 10 \text{ mm thk. leg} & \\ &= 300 \times 19.5 \times 10 = 58.5 \text{ kN}. \end{aligned}$$

② \therefore Rivet value = 29.86 kN.

③ No. of Rivets reqd = $\frac{97}{29.86} = 3.24 \Rightarrow 4 \text{ Nos.}$

④ Pitch = $2.5 \times 19.5 = 48.75 \Rightarrow 55 \text{ mm}$.

⑤ Edge dist = $2 \times 19.5 \Rightarrow 40 \text{ mm}$.



b) Design of welded end connection:

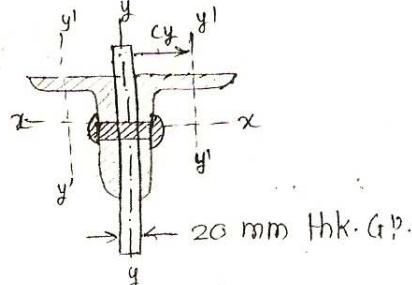
① Max. size of fillet weld = $3/4 \times 8 = 6 \text{ mm}$.

② Length of weld reqd = $\frac{97000}{0.7 \times 6 \times 108} = 213.8 \Rightarrow 215 \text{ mm}$

see problem on Tension member & provide length of weld such that C.G. of weld & c.g. of load coincides. So that no eccentricity will exists.

Pro 3] A strut in a roof truss carries an axial comp load of 180 kN. Determine - a) Design a suitable double angle section for the compression member. The length of strut betn c/c of intersection is 2.3 m & yield stress of steel is 250 MPa.

Soln:- Assume that double angle section is to be connected on both sides of 20 mm thick GP by more than one rivet at each end.



$$\therefore le = 0.85l = 0.85 \times 2.3 = 1.955 \text{ m}$$

① Assume $\delta_{ac} = 80 \text{ MPa}$

$$\therefore A_{reqd} = \frac{180 \times 10^3}{80} = 2250 \text{ mm}^2$$

② Try 2ISA 7750.10 mm,

for each angle, $a = 11.52 \text{ cm}^2$, $r_x = 2.33 \text{ cm}$, $I_y = 21.8 \text{ cm}^4$, $c_y = 1.36 \text{ cm}$.

③ For strut, $a = 2 \times 11.52 = 23.04 \text{ cm}^2$

$$I_y = 2 \times 21.8 + 2 \times 11.52 (1.36 + 1.0)^2 = 171.9 \text{ cm}^4$$

$$\therefore r_y = \sqrt{\frac{I_y}{a}} = \sqrt{\frac{171.9}{23.04}} = 2.73 \text{ cm}$$

$$\therefore r_{min} = 2.33 \text{ cm} = r_x$$

$$④ A_{max} = \frac{l}{r} = \frac{1955}{2.73} = 72.6 \text{ kN} \quad \therefore O.K.$$

⑤ From Table 5.1, $\delta_{ac} = 96.71 \text{ MPa}$

$$⑥ \therefore Safe load = \frac{96.71}{1000} \times 23.04 \times 100 = 222.8 \text{ kN} >> 180 \text{ kN. } \therefore O.K.$$

* End connections :-

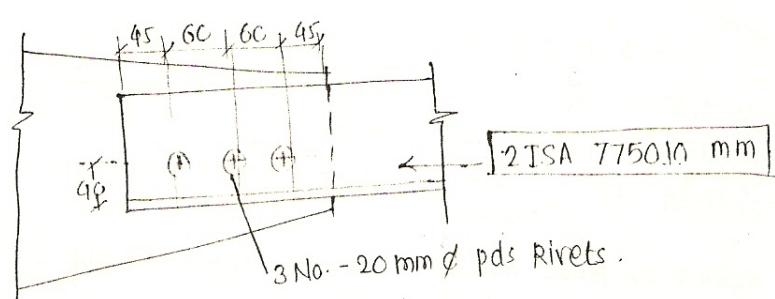
① Rivet value for 20mm dia. pds rivets < Double shear = $2 \times 100 \times \frac{\pi}{4} \times 21.5^2$
 $= 72.6 \text{ kN}$
 Bearing = $300 \times 20 \times 21.5$
 $= 129 \text{ kN}$

$$\therefore \text{Rivet value} = 72.6 \text{ kN.}$$

$$② \text{No. of Rivets reqd} = \frac{180}{72.6} = 2.4 \approx 3 \text{ Nos.}$$

$$③ \text{Pitch} = 60 \text{ mm} \quad (2.5d)$$

$$④ \text{Edge dist} = 45 \text{ mm} \quad (2d)$$



Notes by-

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$$A_{net} = (b - d) t$$


Ag-Hole

$$\begin{aligned} \frac{3A_1}{A_1 + A_2} &= \frac{S_A}{S_A + S_B} \\ A_1 &= (b-d-t/2)t \\ A_2 &= (b-t/2)t \end{aligned}$$

Compression Member

Effective length:-

- ① Both end fixed :- $l_e = 0.8L$
- ② One end fixed :- $l_e = 0.8L$
- ③ S.S. :- $l_e = L$

Capacity of column:-

- Find A_{max} & find bac from TS code
- $A_{max} = \frac{l_e}{y_{min}} \rightarrow 0.65L \rightarrow$ Fixed end
 - $A_{max} = \frac{l_e}{y_{min}} = 0.8L \rightarrow$ one end fixed
 - $\frac{l_e}{y_{min}} = 1L \rightarrow$ S.S.

$$\therefore P = bac \times Ag$$

$$\begin{aligned} bac &= 80 \text{ MPa} \\ bac &= 150 \text{ MPa} \end{aligned}$$

Lacing \Rightarrow Eccentric
Battering \Rightarrow Axial

Design of Lacing:-

$$\text{Given :- } P = 1200 \text{ kN}$$

$$L = 8 \text{ m} \quad l_e = \frac{P}{bac} = \frac{1200}{150} = 10.9 \times 10^3 \text{ mm}^2$$

$$\text{Let } A_{reqd} = \frac{l_e}{bac} = \frac{10.9 \times 10^3}{150} = 72.67 \text{ cm}^2$$

Provide 2 TSLC 350 having
 $Ag = 99.08 \text{ cm}^2$, $y_{min} = 12.72 \text{ cm}$,
 $cy = 60 \text{ mm}$

$$A_{max} = \frac{l_e}{y_{min}} = \frac{10.9 \times 10^3}{12.72} = 85.56$$

Find bac to be assumed
i.e. O.K.

$$\boxed{[S = 200 \text{ mm}]}$$

Let $S = 200 \text{ mm}$

Length of lacing bar = 2L

Assume single lacing system inclined at

45° with axis of column

Length of lacing bar = $2(200 + 2 \times 60)$

= 640 mm

Assume thickness of lacing bar = 12 mm

Assume width of lacing bar = 60 mm

Assume thickness of lacing = 60 mm

Assume width of lacing = 60 mm

∴ $P = bac \times Ag$

$$\begin{aligned} bac &= 110 \text{ MPa} \Rightarrow \text{Lacing} \\ bac &= 150 \text{ MPa} \end{aligned}$$

$$\text{Given :- } P = 1200 \text{ kN}$$

$$l_e = \frac{P}{bac} = \frac{1200}{150} = 8 \text{ m}$$

$$A_{reqd} = \frac{l_e}{bac} = \frac{8}{150} = 53.33 \text{ cm}^2$$

$$A_{max} = \frac{l_e}{y_{min}} = \frac{8}{12} = 66.67 \text{ cm}^2$$

$$A_{reqd} = \frac{53.33}{66.67} = 0.8 \text{ t.f.d.t}$$

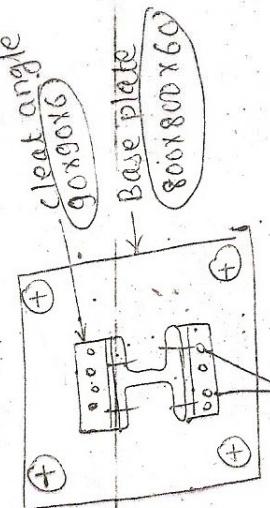
$$A_{reqd} = \frac{53.33}{137.2} = 0.39 \text{ t.f.d.t}$$

Steps in Design:-

- ① $A_{reqd} = \frac{P}{bac} = \frac{P}{110 \text{ MPa}}$
- ② choose section
- ③ check for $A_{max} = \frac{l_e}{y_{min}}$ \rightarrow O.K.
- ④ find $bac > 110 \text{ MPa}$ \rightarrow Assume $bac = 150 \text{ MPa}$
- ⑤ Assume $S = 200 \text{ mm}$
- ⑥ $bac = 45^\circ$ (single lacing) \rightarrow Assume $bac = 45^\circ$
- ⑦ Length of lacing bar = $2(S + 2 \times 60)$
- ⑧ Assume thickness of lacing = 12 mm
- ⑨ Assume width of lacing = 60 mm
- ⑩ Let $N \neq F = \frac{V}{2 \sin \theta}$
- ⑪ Tensile stress = $\frac{F}{A_{net}} = \frac{F}{(b-d) \times t}$
- ⑫ Comp. stress = $\frac{F}{Ag} = \frac{F}{A_g}$ $\neq 110 \text{ MPa}$ or calculated.
- ⑬ $Ag = b \times t$
- ⑭ Provide lacing bar of size $1 \times 60 \times 12 \text{ mm}$
- ⑮ End connection \rightarrow in shearing = $\tau v t \cdot \frac{\pi d^2}{4}$
- ⑯ Rivet value \rightarrow in beginning = $\delta t t \times \delta x t$
- ⑰ No. of Rivet reqd = $\frac{F}{\delta x} = \frac{F}{R}$

$$F_1 + \frac{2}{3}(F - F_1)$$

Slab Base:-



4 Pds 20 mm dia Rivet

$$\text{Area of plate reqd} = \frac{P}{8bc} = 4000$$

$$\therefore \text{Length of plate} = \sqrt{A}$$

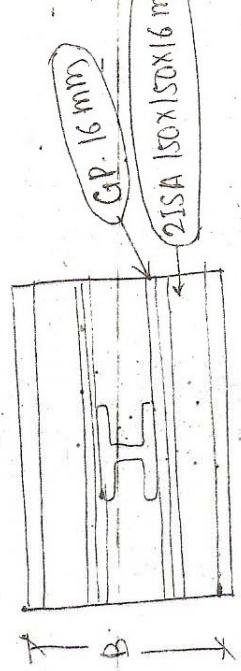
$$\text{thk} = 60 \text{ mm}$$

Cleat angle = 2 ISA 90x90x6
4 - Ø 20 mm Pds Rivet

Concrete slab:-

Area of concrete base = $\frac{1.1 \times P}{\delta_{ac}} = 1000$
Assume 45° dispersion angle.

Crusseted Base



4 Pds 20 mm dia Rivet

$$\text{Area of plate reqd} = \frac{P}{8bc} = 4000$$

Find width = B

$$\text{Length} = \frac{A}{B}$$

To find thk. take section at GP & at angle connection
(t = 20 mm)

$$\text{thk} = 60 \text{ mm}$$

Cleat angle = 2 ISA 90x90x6
4 - Ø 20 mm Pds Rivet

Concrete slab:-

Area of concrete base = $\frac{1.1 \times P}{\delta_{ac}} = 1000$
Assume 45° dispersion angle.

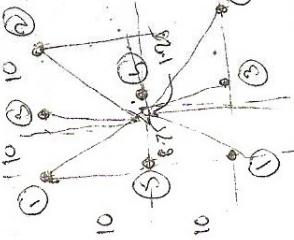
BEAM

Flexural Member

$$\text{find } Z_{\text{reqd}} = \frac{M}{2bc} \rightarrow 150$$

$$\text{Force due to direct load} = F_{\text{c}} = \frac{P}{n} = \frac{60}{2} = 30 \text{ kN}$$

$$\text{Force due to wind} = F_{\text{m}} = \frac{M \cdot x}{2 \cdot l^2} = 25.2$$



$$y = \frac{3 \times 100 + 2 \times 20}{5} = 87.5 \text{ mm}$$

$$Z_{\text{reqd}} = \frac{60 \times 100 \times 116}{60 \times 20 + 2 \times 20 \cos 45^\circ}$$

$$Z_{\text{reqd}} = \frac{60 \times 100 \times 116}{60 \times 20 + 2 \times 20 \cos 45^\circ} = 41.625 \text{ cm}^3$$

$$Q = \frac{60 \times 100}{11.00} = 54.54 \text{ kN}$$

$$\therefore F_c = 31.275 \text{ kN}$$

$$\sigma = \sqrt{I/A}$$

$$\epsilon = \frac{\sigma}{E}$$

Euler formula

$$P_{cr} = \frac{\pi^2 EI}{L^2}$$

for $\frac{\pi^2 E}{L^2}$

$$M = ax^2 + bx + c$$

$$V = 2ax + b$$

$$\therefore W = \frac{dV}{dx} = 2a$$

$$a=2, M=-2$$

$$\text{at } x=0, M=0, \therefore c=0$$

$$\therefore 0=0+0+c \Rightarrow c=0$$

$$-2=4a+2b \quad \text{--- ①}$$

$$-2=4a+2b \quad \text{--- ②}$$

$$\text{at } x=0, V=0 \rightarrow 8$$

$$8=4a+b \quad \text{--- ③}$$

$$a=-3.5$$

$$b=6$$

$$M = -3.5x^2 + 6x + 6$$

$$V = -7x + 6$$

$$W = +7 \text{ kN}$$

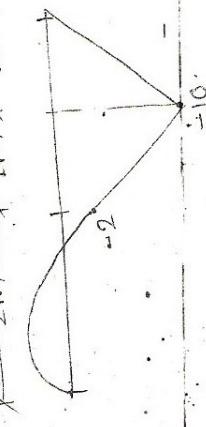
second Mtd:-

$$M = 2RA - 2W - \frac{Wx^2}{2}$$

$$V = 2RA - 2W$$

$$\therefore 2RA - 2W = -2$$

X-axis BMD



Id. S.I.

$$\begin{aligned} \Sigma M &= -2 = 2RA - W \cdot 2 \\ \Rightarrow 2RA - 2W &= -2 \end{aligned}$$

$$\begin{aligned} \text{Parabola} & \\ \text{RA} & \end{aligned}$$

$$\Sigma M = -10 = 3RA - W(2)^2$$

$$\Rightarrow 3RA - 4W = -10$$

$$\therefore RA = 6 \text{ kN/m}$$

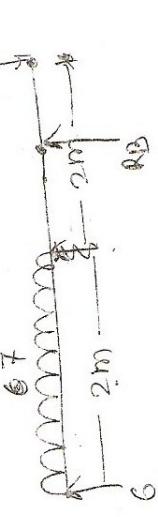
$$\begin{aligned} \text{Parabola} & \\ \text{RA} & \end{aligned}$$

$$\begin{aligned} \Sigma M &= -10 \\ \Rightarrow 3RA - 4W &= -10 \end{aligned}$$

$$\therefore RA = 6 \text{ kN/m}$$

$$W = \frac{7}{4} \text{ kN/m}$$

$$10$$



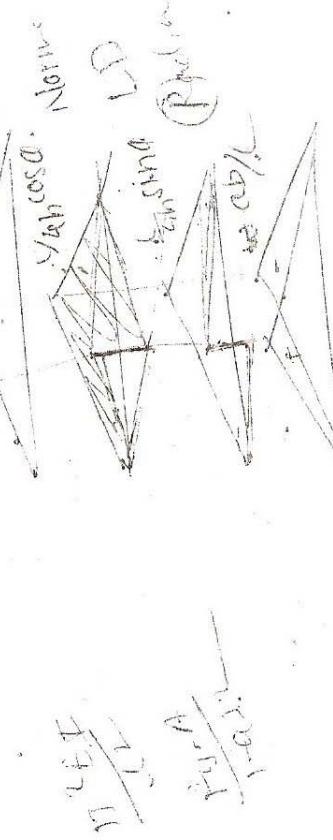
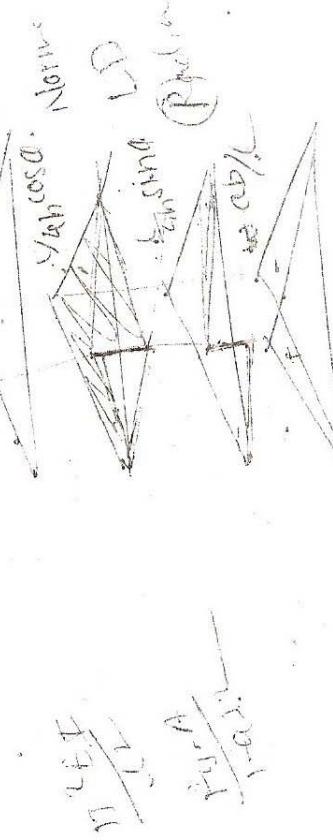
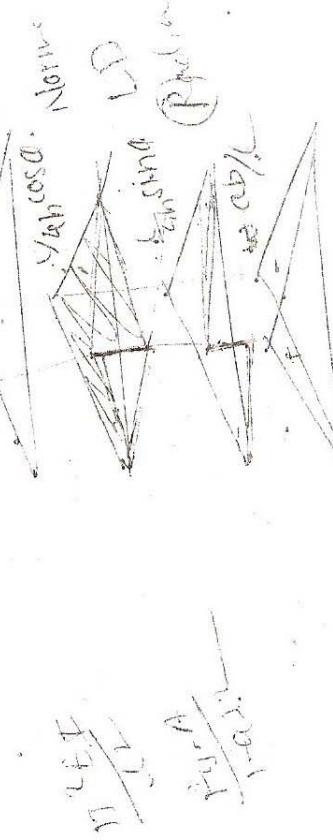
7

2m

5m

10

RA



Rivet:

Strength of joint per pitch =

Smaller of (1)

* Rivet value = In shearing $\tau_{vf} \cdot \frac{\pi}{4} d^2$
In bearing $8pf \cdot d \cdot t$

* Riveted Joint (Axially loaded member)

(1) Dia. of Rivet = $6\sqrt{t \cdot h}$ of plate (mm)

(2) No. of Rivet reqd = $\frac{\text{Load}}{\text{Rivet value}}$

(3) Pitch = 3D

(4) Edge Dist = 2D

	Pds	dia
τ_{vf}	100	80
$8pf$	300	250

* Riveted Joint (Eccentric) :- Bracket I :-

(1) F_a = force due to direct load = $\frac{P}{n}$

(2) F_m = force due to moment = $\frac{P \times R \times M}{2L}$

(3) F_r = Resultant $= \sqrt{F_a^2 + F_m^2 + 2F_a F_m \cos \theta}$

θ = Angle betw F_a & F_m

No tension condition,

$$\frac{\tau_{vf} \text{ cal}}{\tau_{vf}} + \frac{\delta t f \text{ cal}}{\delta t f} \leq 1.4$$

Design steps:-

(1) Assume Rivet dia = 6 $\sqrt{t \cdot h}$ of plate

(2) Rivet value \leftarrow Shear = $\frac{\pi}{4} d^2 \cdot \tau_{vf}$
Bearing = $8pf \cdot d \cdot t$

(3) No. of Rivets reqd = $\frac{P}{R}$

Increase by 3 Times

$$\text{Or } n = \sqrt{\frac{6M}{m \cdot p \cdot R}} \quad \begin{aligned} p &= \text{pitch} \\ m &= \text{No. of rows} \\ R &= \text{Rivet value} \end{aligned}$$

$$(4) \text{ check: } \frac{\tau_{vf} \text{ cal}}{\tau_{vf}} + \frac{\delta t f \text{ cal}}{\delta t f} \leq 1.4$$

$$(5) \delta t f \text{ cal} = \frac{6M}{m \cdot p \cdot n^2 \cdot A_y}$$

(6) $\tau_{vf} \text{ cal} = \frac{\text{Direct shear load on each Rivet}}{\text{Area of Rivet}}$

* Butt Joint Eccentric :-

(1) Shear stress at weld = $P_s = \frac{P}{d \cdot t}$

(2) Tensile/comp. stress due to bending =

$$P_b = \frac{6M}{t \cdot d^2}$$

(3) P_e = Equivalent stress = $\sqrt{P_b^2 + 3P_s^2}$

P_e permissible = $0.9 f_y = 225 \text{ MPa}$

(4) length of weld = $\sqrt{\frac{6M}{E \cdot P_b}}$

$$P_b = 165 \text{ MPa}$$

Fillet weld : Bracket I : Eccentric

(1) Vertical shear stress = $P_s = \frac{W}{L \times t}$

(2) Torsional stress = $P_b = \frac{M \cdot I}{H}$ $I = W \times e$

$$I_p = J \cdot I_y$$

(3) Resultant stress

$$P_r = \sqrt{P_s^2 + P_b^2 + 2P_s P_b \cos \theta}$$

$$P_r \neq 108 \text{ MPa}$$

Fillet weld : Bracket II : Eccentric

(1) Vertical stress = $P_s = \frac{W}{2 \times t}$

(2) Bending stress = $P_b = \frac{6 \times M}{2L}$

(3) Resultant stress = $P_r = \sqrt{P_s^2 + P_b^2} \neq 108 \text{ MPa}$

size of fillet weld = 6 mm

$$\begin{aligned} \text{strength of fillet weld} &= P_f \times L \times 0.707 S \\ &= 108 \times 0.707 \times 5 \times 1 \end{aligned}$$

$$\begin{aligned} A_2 &= (b - t/2) t \\ &= (75 - 4) \times 8 \\ &= 568 \text{ mm}^2 \end{aligned}$$

$$A_{\text{net}} = \frac{3A_1}{3A_1 + A_2} = 0.685$$

$$A_{\text{net}} = A_1 + A_2 K$$

$$= 801.16 \text{ mm}^2$$

$$\begin{aligned} \text{Area of 4 angles} &= 3200 \times 6 \text{ mm}^2 \\ P &= 80t \times A_{\text{net}} \\ &= 150 \times 3200 \times 6 \times 10^{-3} \\ &= 480.7 \text{ kN} \end{aligned}$$

② Tacking is done along A-A, B-B

$$A_1 = (b - d - t/2) t$$

$$= 412 \text{ mm}^2$$

$$K_{\text{Anet}} = \frac{5A_1}{5A_1 + A_2}$$

$$= 0.789$$

$$\begin{aligned} A_{\text{net}} &= A_1 + A_2 K \\ &= 857.23 \text{ mm}^2 \\ P &= 857.23 \times 4 \times 150 \times 10^{-3} \\ &= 514.34 \text{ kN} \end{aligned}$$

- ③ If A & C are tacked.
They acts as double angle section
connected back to back.

A_{net} = Ag - deduction for hole

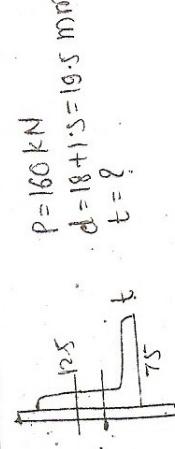
$$= 568 - 195 \times 8 = 982 \text{ mm}^2$$

$$P = 982 \times 150 \times 10^{-3} \times \varphi$$

$$= 589.2 \text{ kN}$$

Tacking is done to all rivet angles.

same as above.



$$P = 160 \text{ kN}$$

$$d = 18 + 1.5 = 19.5 \text{ mm}$$

$$t = ?$$

Design :- ① Anet reqd = $\frac{P}{8at}$ \Rightarrow Pay for section.

$$\text{Find } A_{\text{net}} \text{ reqd} \therefore OK$$

$$\text{② Check of } \lambda := \frac{L}{\sqrt{m}} \text{ min. } \neq 350 \therefore OK$$

$$\text{③ Tension splice:- } \lambda := \frac{L}{\sqrt{m}} \text{ min.}$$

Lug angle :-



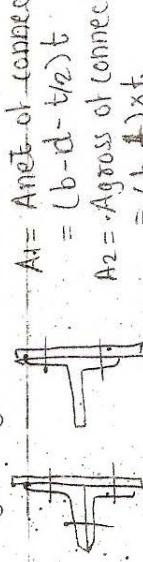
Length of Tension member is less.
To save length of G.F.

- ④ Area of plate : A_{net} = (b - nd - t/2) × t

If joints are staggered,

$$\text{add } \left(\frac{S^2 t}{4q} \right)$$

Double angles on same side of G.P (back to back)



$$\begin{aligned} A_1 &= A_{\text{net}} \text{ of connected leg} \\ &= (b - d - t/2) t \\ A_2 &= A_{\text{gross}} \text{ of connected leg} \\ &= (b - \frac{t}{2}) t \end{aligned}$$

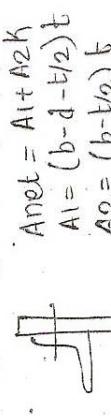
$$A_{\text{net}} = A_1$$

$$K = \frac{5A_1}{3A_1 + A_2}$$

$$A_{\text{net}} = A_1 + A_2 K$$

$$= A_{\text{net}}$$

⑤ Single angle \leftrightarrow connected to G.P



$$A_{\text{net}} = A_1 + A_2 K$$

$$A_1 = (b - d - t/2) t$$

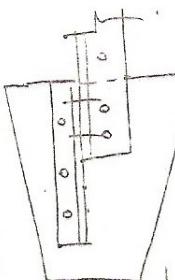
$$A_2 = (b - t/2) t$$

$$A_{\text{net}} = \frac{5A_1}{3A_1 + A_2}$$

$$A_{\text{net}} = 6 A_1 + A_2 K$$

⑥ Double angle on either side of G.P.

Anet = Ag - deduct for hole.



Length of Tension member is less.
To save length of G.F.